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FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CRITICAL HARDWARE

NUMBER: M4-18G-FL030-X

SUBSYSTEM MAME: ELECTRICAL POWER GENERATION - CRYO, GENERIC

REVISION: 1 11/12/91

PART NAME VENDOR NAME

PART NUMBER VENDOR NUMBER

a SRU :

WINTEC ) H 2

MC286-0054-0001

24267-605

 $F_{M}$ 

PART DATA

EXTENDED DESCRIPTION OF PART UNDER ANALYSIS: FILTER, H2

■ REFERENCE DESIGNATORS:

40V45FL030

40V45FL04Q

40V45FL500

40V45FL560

40V45FL660

40V45FL863

40V45FL873

40Y45FL883

: 40V45FL893

■ QUANTITY OF LIKE ITEMS: 1 ONE PER TANK

□ FUNCTION:

PROTECTS THE FCP'S FROM ANY PARTICULATE CONTAMINATION WHICH MAY BE PRESENT IN H2 TANKS.

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15.0

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FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CRITICAL FAILURE MODE NUMBER: M4-18G-FL030-01

REVISION#

1 11/12/91 R

SUBSYSTEM: ELECTRICAL POWER GENERATION - CRYO, GENERIC

ITEM NAME: FILTER , ₩2\_

CRITICALITY OF THIS

FAILURE MODE: 1R2

■ FAILURE MODE:

PLJGGED OR RESTRICTED

MISSION PHASE:

LO LIFT-OFF 00CM-ORBIT 30 DE-ORBIT

LS LANDING SAFING

■ VEHICLE/PAYLOAD/KIT EFFECTIVITY: 102 COLUMBIA

: 103 DISCOVERY : 104 **ATLANTIS** : 105 ENGEAVOUR

■ CAUSE: CONTAMINATION.

- CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO
- REDUNDANCY SCREEN A) PASS

8) PASS- -7

8 C) PASS

PASS/FAIL RATIONALE:

33 A)

a 8)

G C)

## - FAILURE EFFECTS -

- (A) SUBSYSTEM: SUBSYSTEM DEGRACATION - LOSS OF CAPABILITY TO DISTRIBUTE CONSUMABLES FROM ONE H2 SUPPLY TANK.
- (8) INTERFACING SUBSYSTEM(S): REDUCED QUANTITY OF HZ CONSUMABLES AVAILABLE FOR FUEL CELL POWERPLANT OPERATION.

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FAILURE FORES EFFECTS ANALYSIS (FMEA) -- CRITICAL FAILURE MODE

NUMBER: M4-18G-FL030-01

■ (C) MISSION: MINIMUM DURATION MISSION INVOKED.

- (D) CREW. VEHICLE. AND ELEMENT(S): NO EFFECT AFTER FIRST FAILURE.
- (E) FUNCTIONAL CRITICALITY EFFECTS:
  FUNCTIONAL CRITICALITY EFFECTS COMPLETE BLOCKAGE MAY RESULT IN LOSS
  OF CREW/VEHICLE IF COMBINED WITH A FAILED CLOSED TANK RELIEF VALVE
  (TANK RUPTURE).

- DISPOSITION RATIONALE -

- (A) DESIGN:
  DURING DEVELOPMENT TESTS, SUMP CAPACITY INCREASED FROM 0.204 CU IN. TO
  0.944 CU IN. TO ADD TO CONTAMINANT CAPACITY. 12 MICRON ABSOLUTE FILTER.
  ELEMENT REPLACED EVERY 25 MISSIONS OR AS DETERMINED BY SYSTEM CHECKOUT.
- © (8) TEST: QUALIFICATION TEST VERIFIED PRESSURE DROP LESS THAN 15 PSI AFTER INTRODUCTION OF 4 GRAMS OF AC FINE DUST AT MAXIMUM SYSTEM FLOW RATES.

ACCEPTANCE TEST VERIFIES LEVEL 200A CLEANLINESS BY PARTICLE COUNT AND NON-VOLATILE RESIDUE (MVR), BUBBLE POINT AND PRESSURE DROP ARE WITHIN TOLERANCE. DURING PANEL MODULAR ASSEMBLY, FILTER PERFORMANCE IS FURTHER VERIFIED.

CMRSO: PRSD FILTER FLOW VERIFIED IN-FLIGHT.

(C) INSPECTION: RECEIVING INSPECTION MATERIAL CERTIFICATIONS AND TEST REPORTS ARE REVIEWED TO VERIFY THAT MATERIAL COMPOSITION, DIMENSIONS, AND SURFACE CONDITIONS COMPLY WITH DESIGN REQUIREMENTS.

CONTAMINATION CONTROL

ACCOMPLISHMENT OF PREASSEMBLY CLEANING, VAPOR DEGREASING, AND
ULTRASONIC CLEANING IS VERIFIED. PIECE PART CLEANLINESS IS CERTIFIED
TO LEVEL 200A (MA0110-301) BY A NVR AND PARTICLE COUNT (50 ML FLUSH
THROUGH A MILLIPORE FILTER); PIECE PARTS ARE INSPECTED FOR BURRS.
COMPONENT ASSEMBLY, PACKAGING, AND SUBSEQUENT SUBASSEMBLY INSTALLATION
ARE ACCOMPLISHED IN 100,000 CLASS CLEANROOM ENVIRONMENTS. COMPLETED
ASSEMBLY IS VERIFIED TO LEVEL 200A.

ASSEMBLY/INSTALLATION

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FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CRITICAL FAILURE MODE
NUMBER: K4-18G-FL030-01

NO PALCGEMATED OR SULFONATED SOLVENTS OR METHYL ALCOHOL ALLOWED DURING ANY PROCESSING OF MESH PACK OR ELEMENT ASSEMBLY.

TESTING ATP VERIFIES LEVEL 200A CLEANLINESS.

HANDLING/PACKAGING
PACKAGING IS INSPECTED FOR INTEGRETY AND COMPLIANCES WITH MATERIAL FANDLING REQUIREMENTS.

- (D) FAILURE HISTORY:
  THERE HAVE BEEN NO ACCEPTANCE TEST, QUALIFICATION FEST, FIELD OR FLIGHT
  FAILURES ASSOCIATED WITH THIS FAILURE MODE.
- (E) OPERATIONAL USE:

  REDUNDANT TANK HEATERS ENABLED TO SUPPLY REACTANTS TO THE FUEL CELLS.

  FOR A COMPLETELY PLUGGED FILTER, CREW WILL DEACTIVATE ASSOCIATED TANK

  HEATERS IN RESPONSE TO HIGH PRESSURE FAULT ANNUNCIATION.

## - APPROVALS -

RELIABILITY ENGINEERING: M. O. WEST
DESIGN ENGINEERING : M. M. SCHEIERN
QUALITY MANAGER : O. J. BUTTNER
NASA DELIABILITY

MASA RELIABILITY
MASA SUBSYSTEM MANAGER:
MASA QUALITY ASSURANCE: